

# A multirate variational approach to simulation and optimal control for flexible spacecraft

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# Outline

- Short overview
- Mathematical model
- Multirate DMOC
- Computational results and discussion

- **Flexible spacecraft**

- Stringent performance and positioning requirements
- Efficient operation
- Lightweight structure designs and induced vibrations- potentially degrading the performance, causing loss of pointing accuracy or even structural damage

-> Need for efficient control method maximizing the system's performance while respecting all hard safety-critical constraints

- **Multirate Discrete Mechanics and Optimal Control (Multirate DMOC) <sup>1-4</sup> :**

- ✓ **High fidelity simulation at a reduced computational cost**

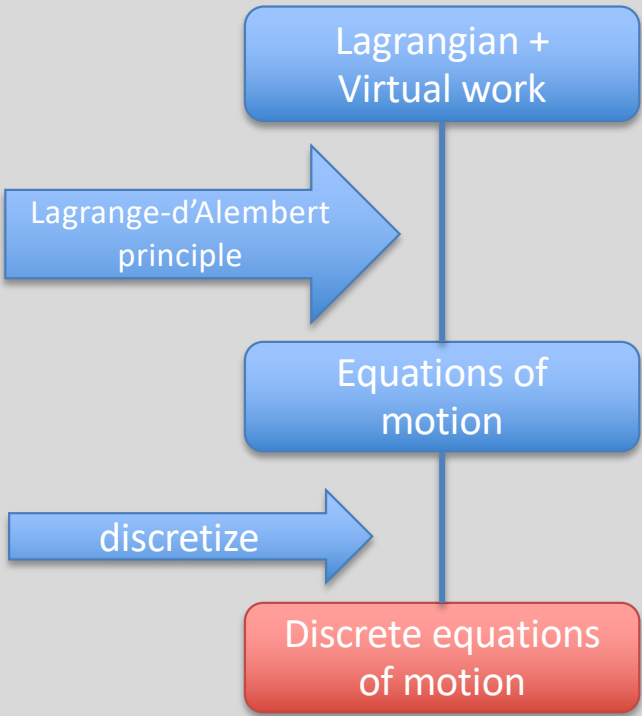
# Advantages

- **Multirate Discrete Mechanics and Optimal Control (Multirate DMOC):**

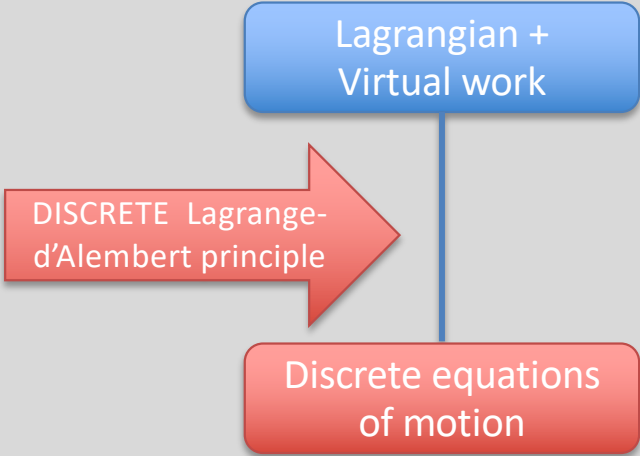
High fidelity simulation at a reduced computational cost

- ✓ Structure-preserving -> *conservation of energy and/or momenta of the system*

### Standard direct methods



### DMOC

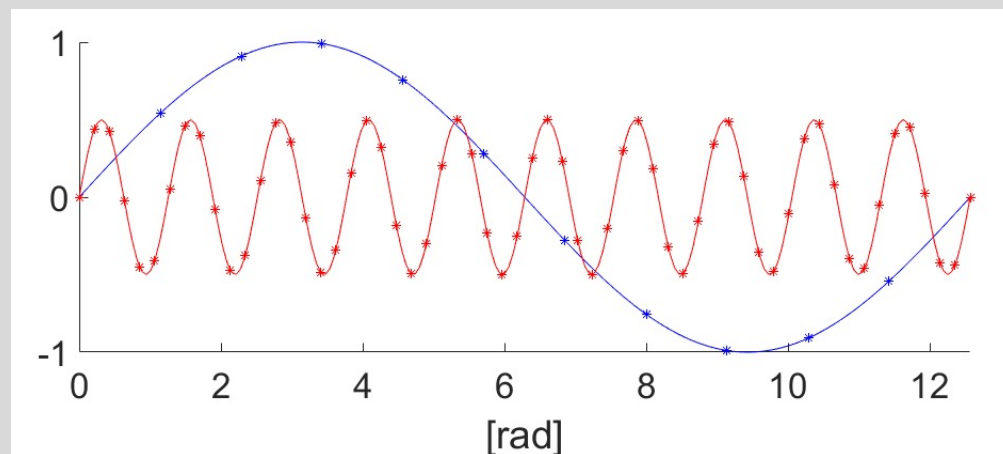
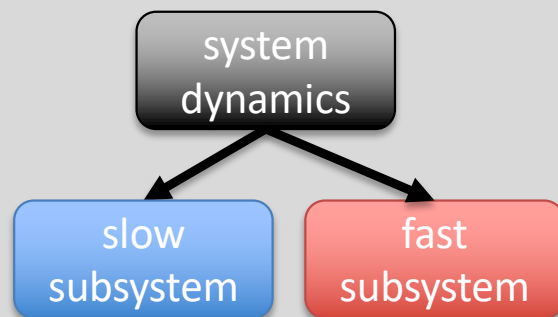


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## High fidelity simulation at a reduced computational cost

- ✓ Structure-preserving -> *conservation of energy and/or momenta of the system*
- ✓ **Multirate discretization -> *computational efficiency***<sup>5</sup>



- **Reduces the number of optimization variables**
- **Reduces the number of equality constraints**

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- ✓ **Unified control methodology -> *potential improvement in optimality and constraint-handling capabilities\****

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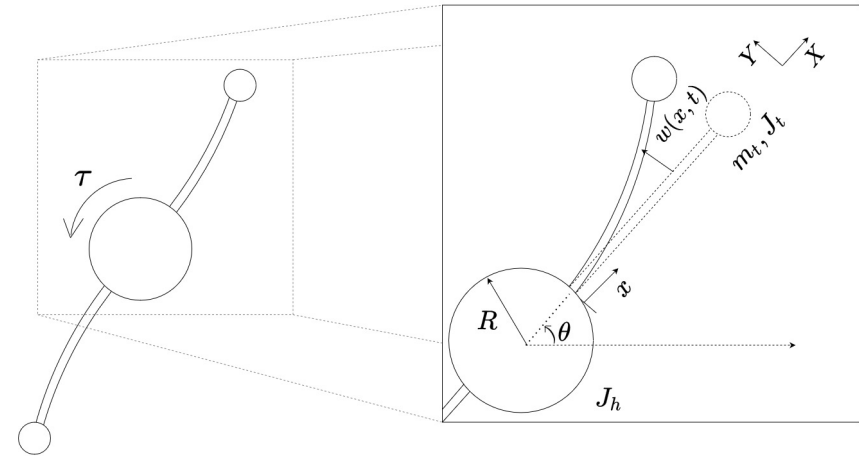
- **General linear model neglecting dissipation effects**
  - single-axis rest-to-rest rotational maneuver
  - control torque applied at the hub

- **Spatial discretization - The Assumed Modes Method<sup>7</sup>**

$$w(x, t) = \sum_{j=1}^N \phi_j(x) \eta_j(t), \quad x \in [0, L]$$

$\phi_j$  - assumed spatial mode shapes  
 $\eta_j$  - time varying modal amplitudes

$N$  - number of modes retained in the approximation  
 $L$  - length of the beam



- **System description<sup>7</sup>**

$$\underline{\xi} = \begin{bmatrix} \theta \\ \underline{\eta} \end{bmatrix}, \quad \underline{\eta} = [\eta_1, \eta_2, \dots, \eta_N]^T \in R^{N \times 1}$$

- Lagrange - d'Alembert principle

$$\mathcal{L}(\underline{\xi}, \dot{\underline{\xi}}) = \frac{1}{2} \dot{\underline{\xi}}^T \mathbf{M} \dot{\underline{\xi}} - \frac{1}{2} \underline{\xi}^T \mathbf{K} \underline{\xi}, \quad \delta W = \underline{f} \cdot \delta \underline{\xi} = \tau \delta \theta$$

$$\delta \int_{t_0}^{t_f} \mathcal{L}(\underline{\xi}, \dot{\underline{\xi}}) dt + \int_{t_0}^{t_f} (\underline{f} \cdot \delta \underline{\xi}) dt = 0 \text{ for all variations } \delta \underline{\xi} \text{ with } \delta \underline{\xi}(t_0) = \delta \underline{\xi}(t_f) = 0$$

$$\mathbf{M} \ddot{\underline{\xi}} + \mathbf{K} \underline{\xi} = \mathbf{D} \tau, \quad \mathbf{M} = \begin{bmatrix} M_{\theta\theta} & M_{\theta\eta}^T \\ M_{\theta\eta} & M_{\eta\eta} \end{bmatrix}, \quad \mathbf{K} = \begin{bmatrix} 0 & 0 \\ 0 & K_{\eta\eta} \end{bmatrix}, \quad \mathbf{D} = [1, 0, \dots, 0]^T$$

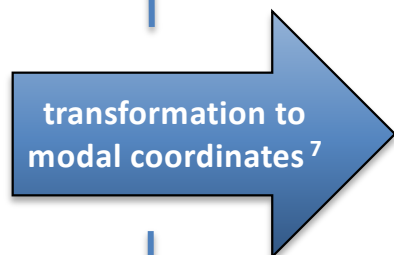
# Transformation to modal coordinates

$$\underline{\xi} = \begin{bmatrix} \theta \\ \eta \end{bmatrix}$$

$$\mathcal{L}(\underline{\xi}, \dot{\underline{\xi}}) = \frac{1}{2} \dot{\underline{\xi}}^T \mathbf{M} \dot{\underline{\xi}} - \frac{1}{2} \underline{\xi}^T \mathbf{K} \underline{\xi}$$

$$\underline{f} = \mathbf{D} \tau$$

$$\mathbf{M} \ddot{\underline{\xi}} + \mathbf{K} \underline{\xi} = \mathbf{D} \tau$$



$$\underline{\xi} = \mathbf{E} \underline{q}$$

$$\mathcal{L} = \frac{1}{2} (\dot{\underline{q}}^T \dot{\underline{q}} - \underline{q}^T \mathbf{\Lambda} \underline{q})$$

$$\underline{f} = \mathbf{E}^T \mathbf{D} \tau$$

$$\ddot{\underline{q}} + \mathbf{\Lambda} \underline{q} = \mathbf{E}^T \mathbf{D} \tau$$

$$\mathbf{E}^T \mathbf{M} \mathbf{E} = \mathbf{I}, \quad \mathbf{E}^T \mathbf{K} \mathbf{E} = \mathbf{\Lambda},$$

$$w_1 \leq w_2 \leq \dots \leq w_{N+1}, \quad \mathbf{\Lambda} = \begin{bmatrix} \omega_1^2 & \dots & 0 \\ \vdots & \ddots & \vdots \\ 0 & \dots & \omega_{N+1}^2 \end{bmatrix}$$

**Table 1:** Structural parameters used for the simulations

Hub radius	$R$	1.0	ft
Hub rotary inertia	$J_h$	8.0	slug-ft <sup>2</sup>
Tip mass	$m_t$	0.156941	slug
Tip mass rotary inertia	$J_t$	0.0018	slug-ft <sup>2</sup>
Beam length	$L$	4.0	ft
Beam height	$h$	6.0	in.
Beam thickness	$t$	0.125	in.
Beam linear density	$\rho A$	0.0271875	slug/ft
Beam elastic modulus	$E$	$0.1584 \times 10^{10}$	lb/ft <sup>2</sup>

**Table 2:** Natural frequencies

$w_1$	0	rad/s
$w_2$	6.454	rad/s
$w_3$	52.41	rad/s
$w_4$	$1.607 \times 10^2$	rad/s
$w_5$	$3.381 \times 10^2$	rad/s
$w_6$	$5.78 \times 10^2$	rad/s

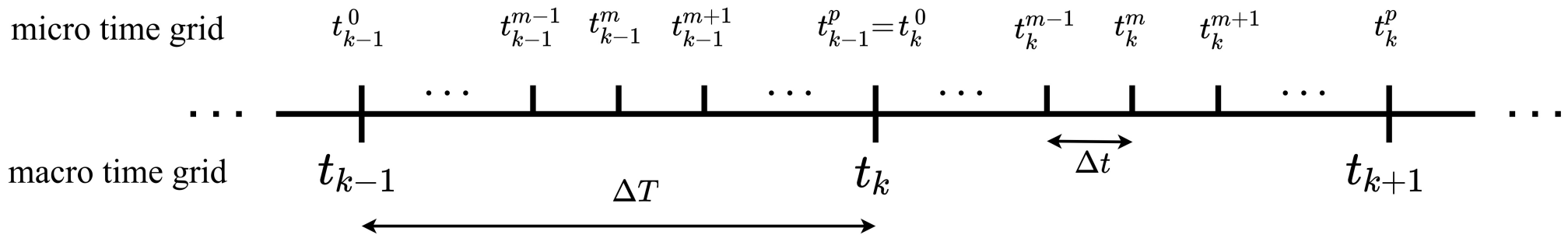
$$N = 5$$

# Multirate formulation

$$\underline{q} = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \\ q_5 \\ q_6 \end{bmatrix} \left. \begin{array}{l} \} \\ \} \end{array} \right\} \begin{array}{l} \underline{q}_s \\ \underline{q}_f \end{array} \longrightarrow \begin{array}{l} w_1 = 0 \text{ rad/s} \\ w_2 = 6.454 \text{ rad/s} \\ w_3 = 52.41 \text{ rad/s} \\ w_4 = 1.607 \times 10^2 \text{ rad/s} \\ w_5 = 3.381 \times 10^2 \text{ rad/s} \\ w_6 = 5.78 \times 10^2 \text{ rad/s} \end{array}$$

$$\mathcal{L} = \frac{1}{2} \left( (\underline{\dot{q}}^s)^T \underline{\dot{q}}^s - (\underline{q}^s)^T \mathbf{\Lambda}_s \underline{q}^s \right) + \frac{1}{2} \left( (\underline{\dot{q}}^f)^T \underline{\dot{q}}^f - (\underline{q}^f)^T \mathbf{\Lambda}_f \underline{q}^f \right)$$

$$\text{where } \mathbf{\Lambda}_s = \text{diag}(\omega_1^2, \dots, \omega_3^2), \mathbf{\Lambda}_f = \text{diag}(\omega_4^2, \dots, \omega_6^2)$$



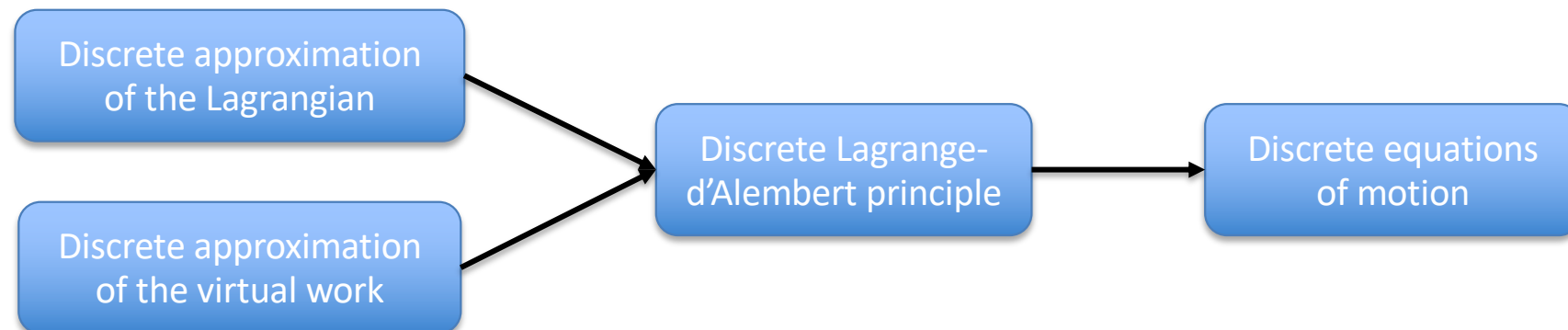
$$p = \frac{\text{macro time step}}{\text{micro time step}} = \frac{\Delta T}{\Delta t}$$

# Multirate formulation

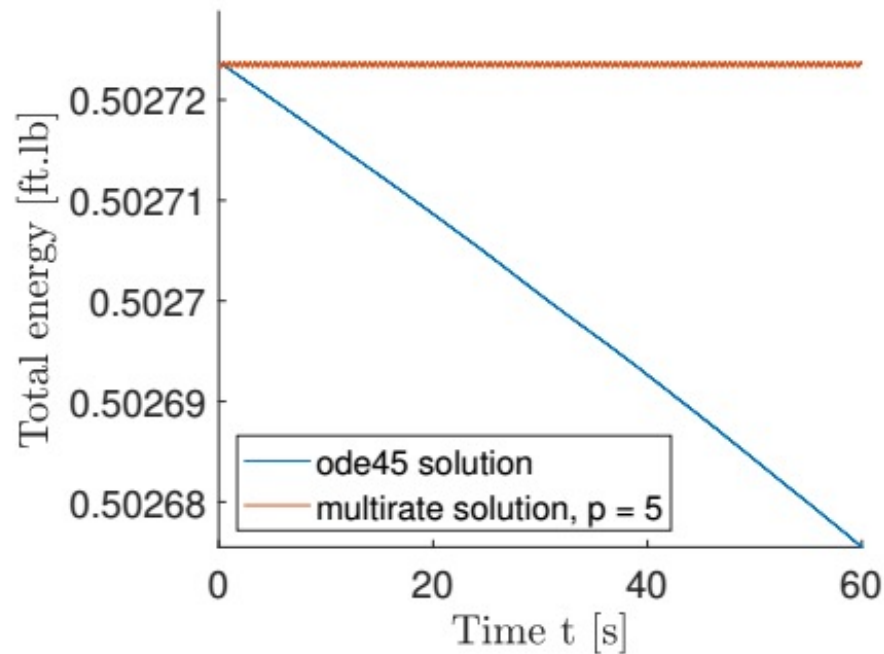
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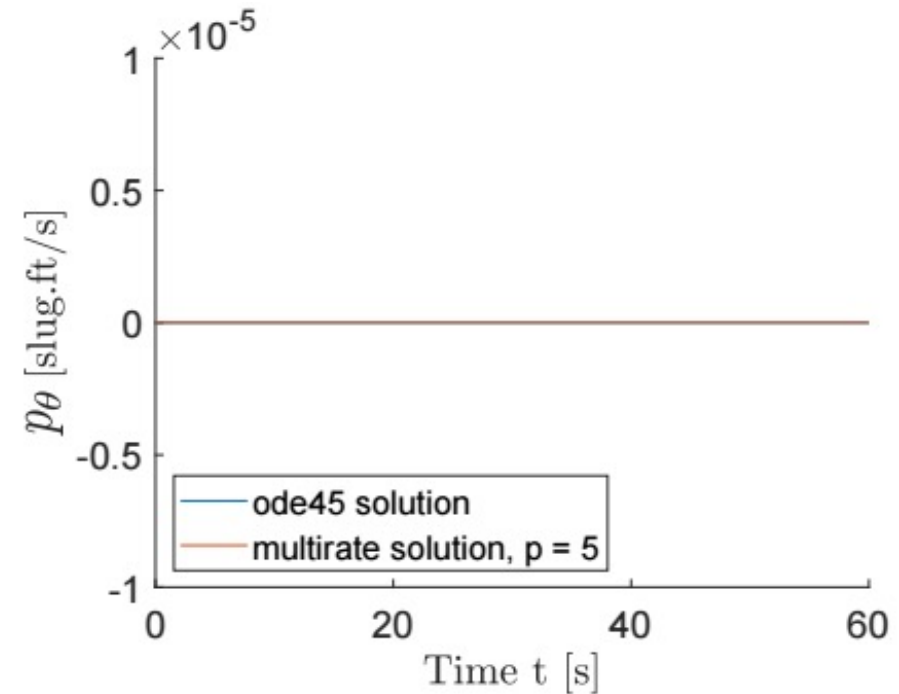
where  $\mathbf{\Lambda}_s = \text{diag}(\omega_1^2, \dots, \omega_3^2)$ ,  $\mathbf{\Lambda}_f = \text{diag}(\omega_4^2, \dots, \omega_6^2)$



# Forward simulation without control



**Figure 1:** Numerical dissipation of total energy for simulations with  $\Delta t = 10^{-4}$  and  $t_f = 60$ s



**Figure 2:** Momentum preservation for simulations with  $\Delta t = 10^{-4}$  and  $t_f = 60$ s

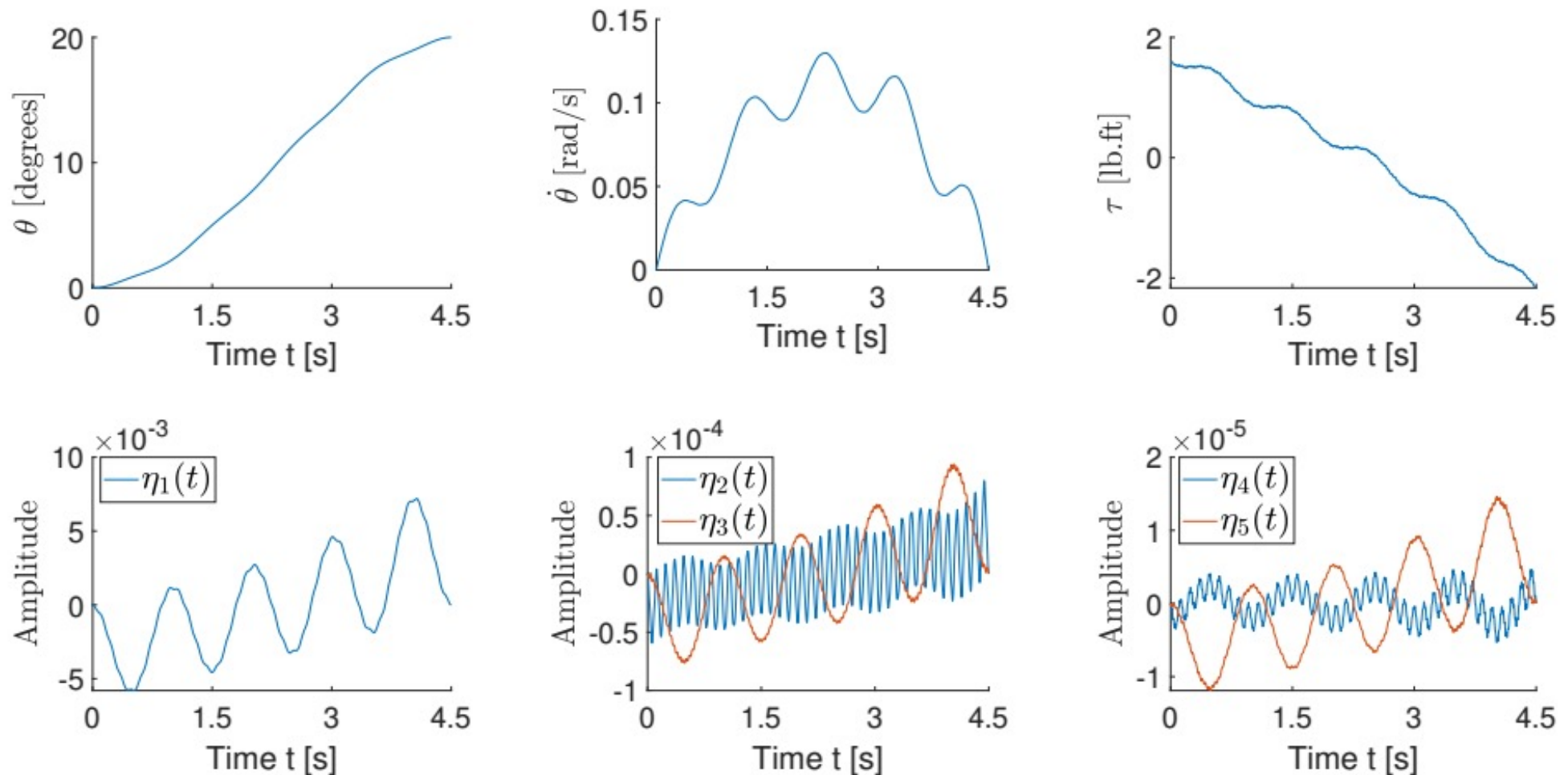
$$p_\theta = \frac{\partial \mathcal{L}}{\partial \dot{\theta}} = M_{\theta\theta} \dot{\theta} + (M_{\theta\eta})^T \underline{\dot{\eta}}$$

$$J(x, u) = \int_{t_0}^{t_f} C(\underline{x}(t), u(t)) dt = \frac{1}{2} \int_{t_0}^{t_f} [\underline{x}(t)^T \mathbf{W} \underline{x}(t) + u(t)^2] dt$$

$$\text{subject to } \left\{ \begin{array}{l} \dot{\underline{x}}(t) = \mathbf{A} \underline{x}(t) + \mathbf{B} \tau(t) \\ \underline{q}(t_0) = \mathbf{E}^{-1} \underline{\xi}_{t_0}, \quad \underline{\xi}_{t_0} = [0, \dots, 0]^T \\ \underline{q}(t_f) = \mathbf{E}^{-1} \underline{\xi}_{t_f}, \quad \underline{\xi}_{t_f} = [\theta_{t_f}, 0, \dots, 0]^T \\ \underline{\dot{q}}(t_0) = \underline{\dot{q}}(t_f) = [0, \dots, 0]^T \end{array} \right.$$

$$\text{where } \underline{x}(t) = \begin{bmatrix} \underline{q}(t) \\ \underline{\dot{q}}(t) \end{bmatrix}, \quad u(t) = \tau(t), \quad \mathbf{A} = \begin{bmatrix} 0 & \mathbf{I} \\ -\mathbf{\Lambda} & 0 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} 0 \\ \mathbf{E}^T \mathbf{D} \end{bmatrix}, \quad \mathbf{W} = \mathbf{I}$$

# Example solution

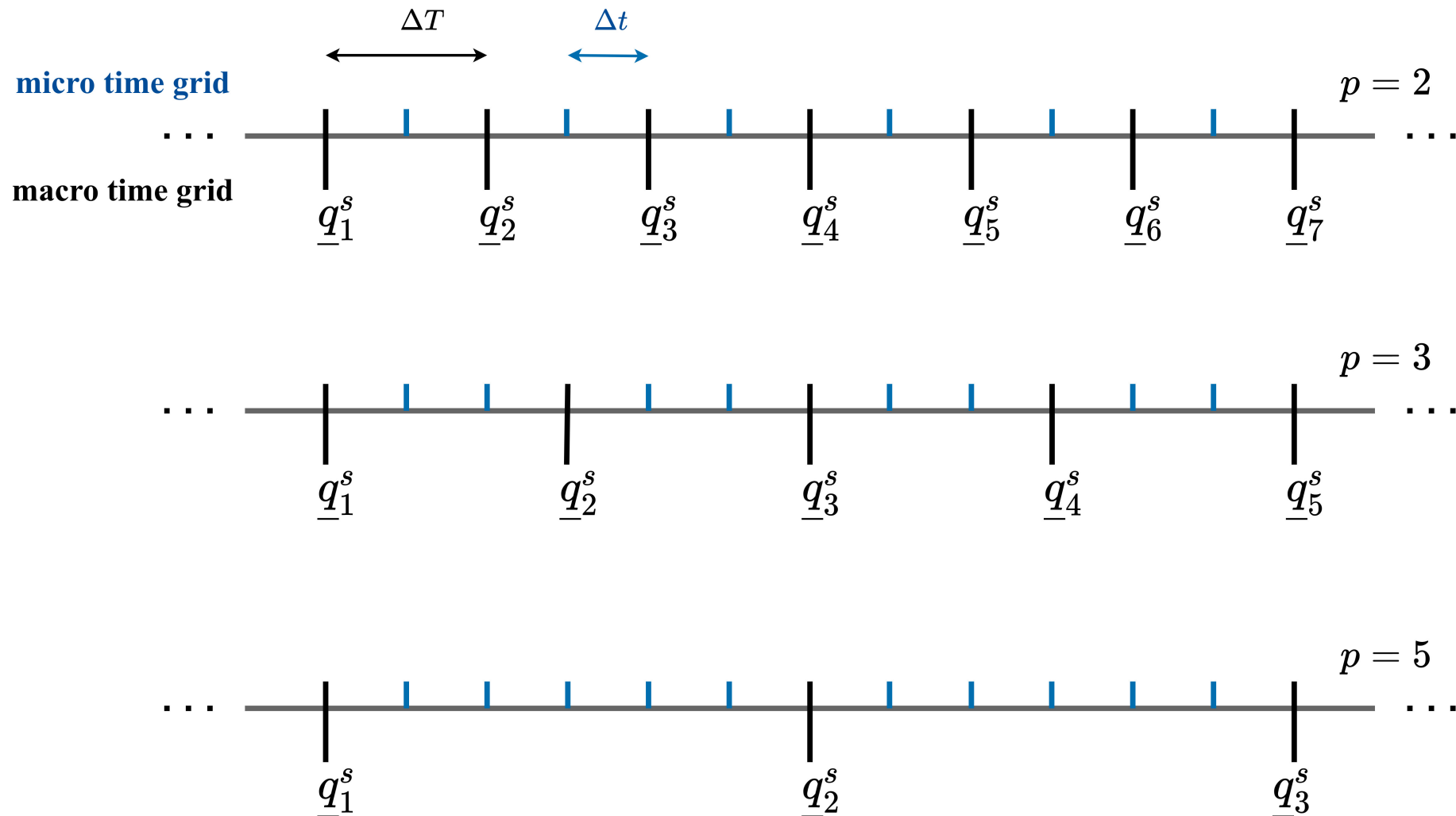


**Figure 3:** OCP solution with Multirate DMOC for  $\theta_{t_f} = 20^\circ$ ,  $t_f = 4.5$  s,  $\Delta t = 10^{-3}$  and  $p = 5$



# Problem size

$$p = \frac{\text{macro time step}}{\text{micro time step}} = \frac{\Delta T}{\Delta t}$$



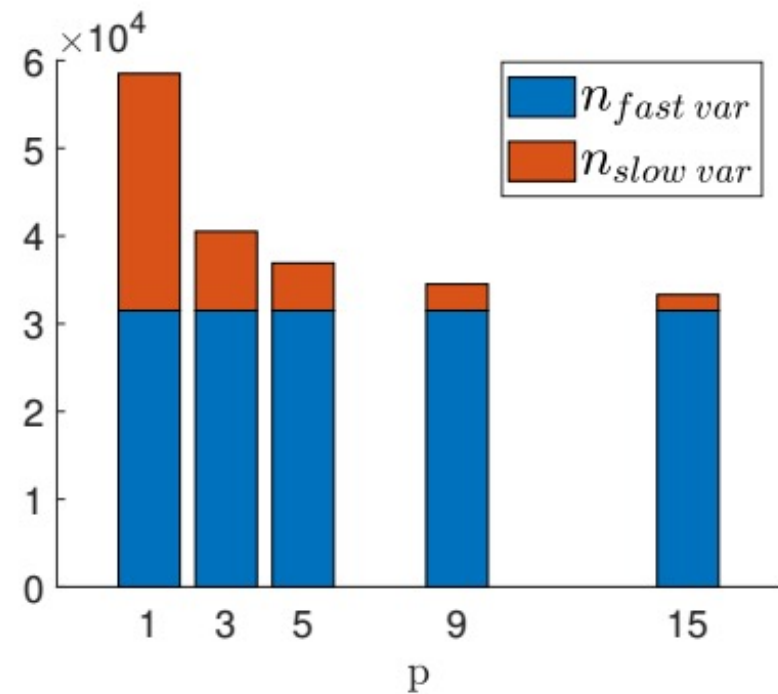
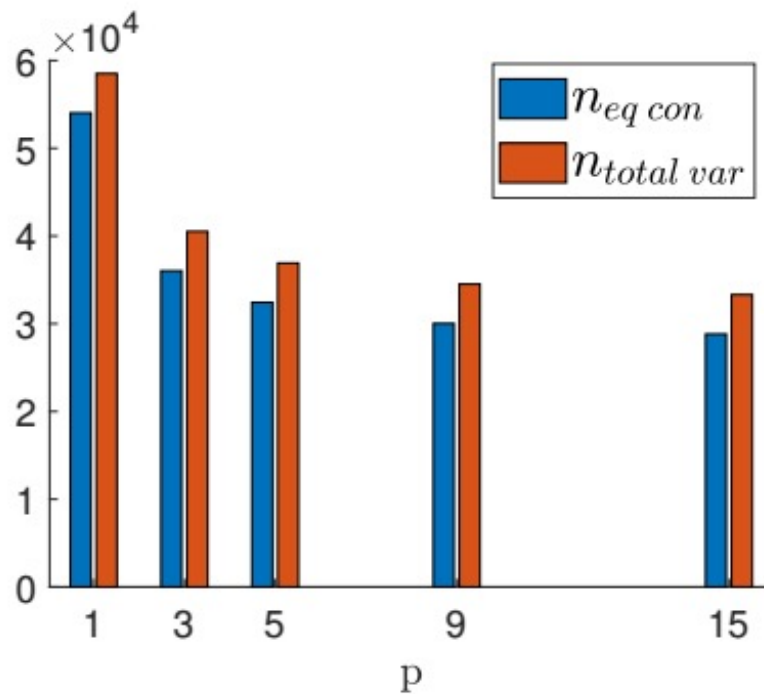
# Problem size

$n_{eq\ con}$  - number of equality constraints

$n_{slow\ var}$  - number of optimization variables resulting from discretization on the macro grid

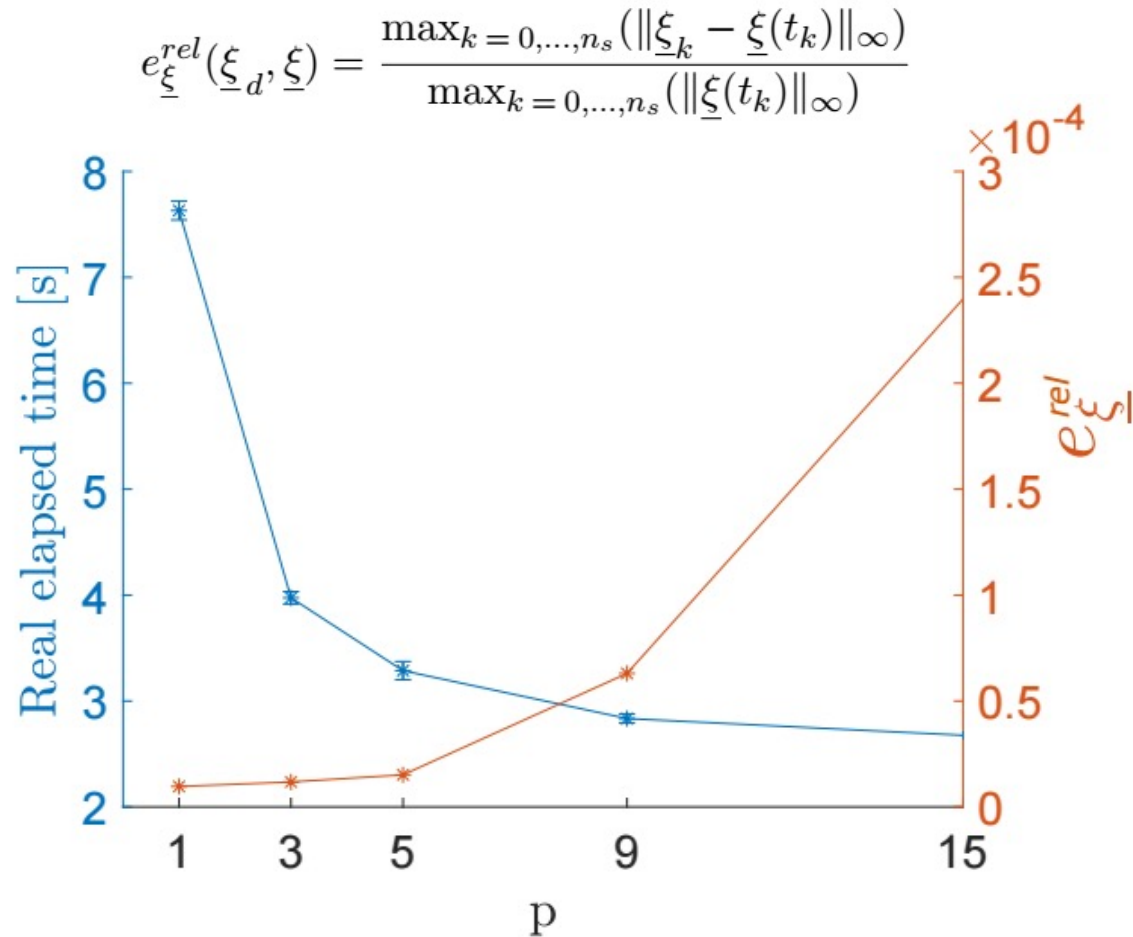
$n_{fast\ var}$  - number of optimization variables resulting from discretization on the micro grid

$$n_{total\ var} = n_{slow\ var} + n_{fast\ var}$$



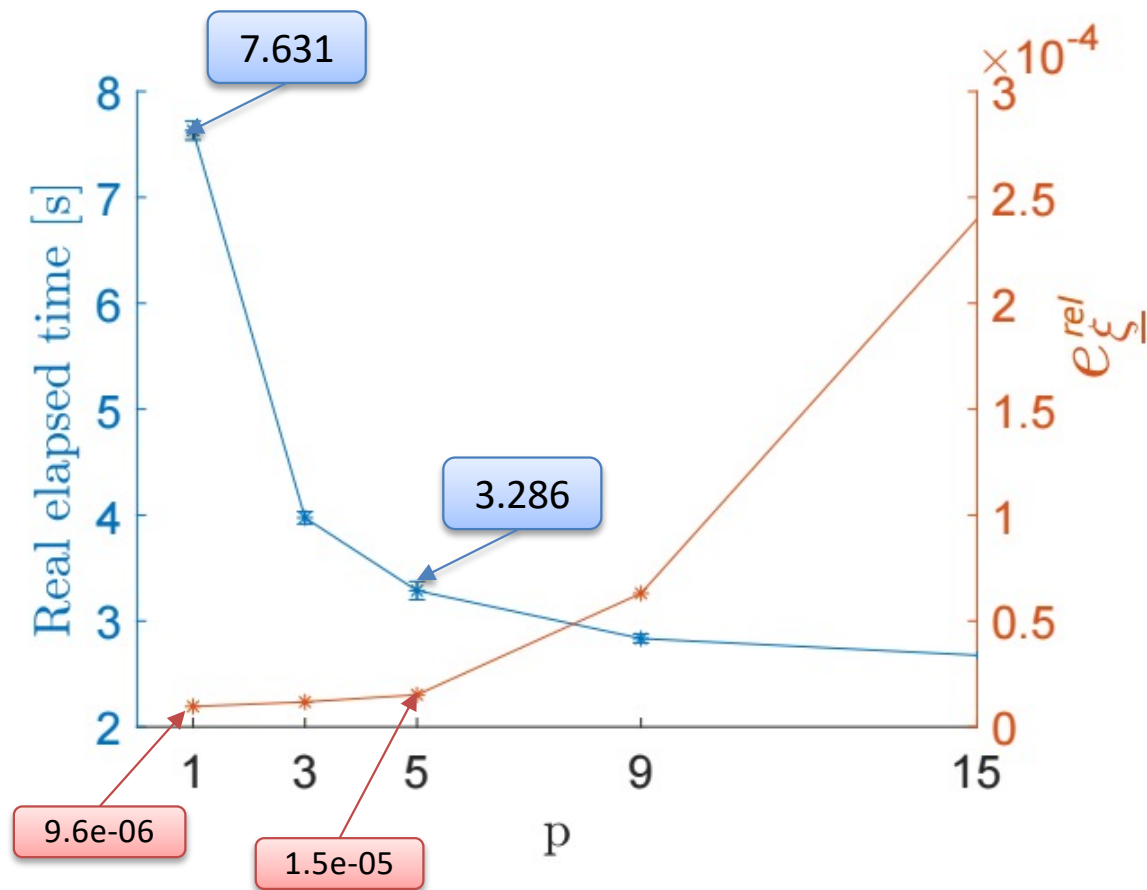
**Figure 4:** Size of OCP based on Multirate DMOC for a simulations with  $\Delta t = 10^{-3}$  and  $t_f = 4.5s$

# Main result- the trade-off



**Figure 5:** Mean computational time with standard deviation and relative error in  $\underline{\xi}$  versus  $p$  for a constant micro time step of  $10^{-3}$ ,  $t_f = 4.5s$  and  $\theta_{t_f} = 20^\circ$

# Main result- the trade-off



**Figure 5:** Mean computational time with standard deviation and relative error in  $\underline{\xi}$  versus  $p$  for a constant micro time step of  $10^{-3}$ ,  $t_f = 4.5s$  and  $\theta_{t_f} = 20^\circ$

# Further customization

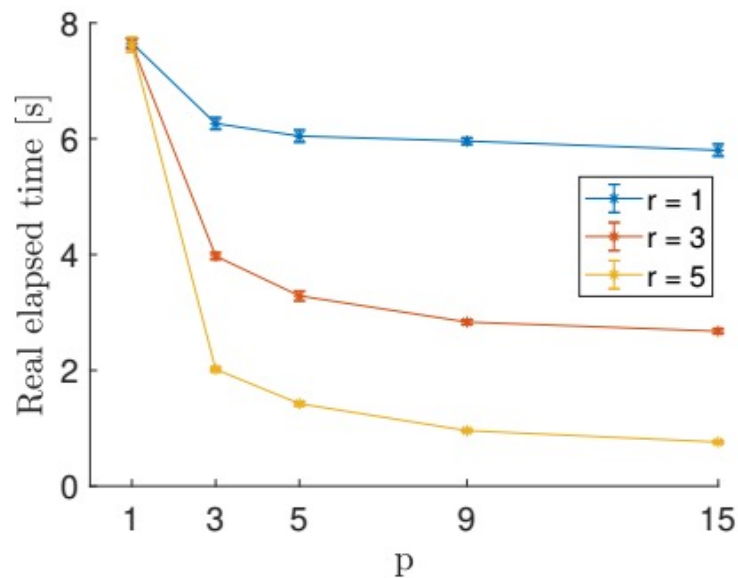
$$\underline{q}^s \in R^{r \times 1} \quad r - \text{degrees of freedom of the slow subsystem}$$

$$\underline{q}^f \in R^{(N+1-r) \times 1} \quad \text{so far in the examples } r = 3, N = 5$$

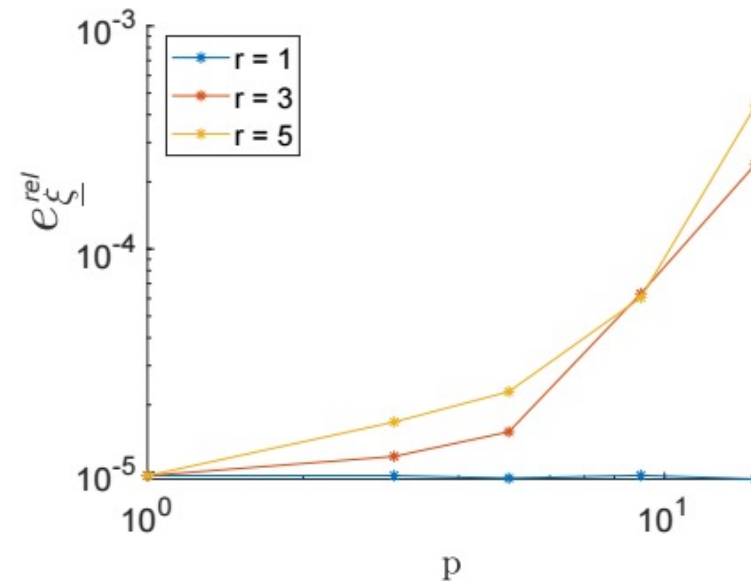
$$n_{total \text{ var}} = n_{slow \text{ var}} + n_{fast \text{ var}}$$

$$n_{slow \text{ var}}(p, r, N, t_f, \Delta t) = 2r \left( \frac{t_f}{p \Delta t} + 1 \right)$$

$$n_{fast \text{ var}}(r, N, t_f, \Delta t) = 2(N + 1 - r) \left( \frac{t_f}{\Delta t} + 1 \right) + \frac{t_f}{\Delta t}$$



**Figure 6:** Mean computational time with standard deviation versus  $p$  for a constant micro time step of  $10^{-3}$ ,  $t_f = 4.5s$  and  $\theta_{t_f} = 20^\circ$



**Figure 7:** Relative error in  $\xi$  versus  $p$  for a constant micro time step of  $10^{-3}$ ,  $t_f = 4.5s$  and  $\theta_{t_f} = 20^\circ$

# Conclusions

- **Multirate Discrete Mechanics and Optimal Control (Multirate DMOC):**

## High fidelity simulation at a reduced computational cost

- ✓ Structure-preserving -> *conservation of energy and/or momenta of the system*
- ✓ Multirate discretization -> *computational efficiency*
- ✓ Unified control methodology -> *potential improvement in optimality and constraint- handling capabilities*
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- ✓ Straightforward selection of slow and fast subsystems -> *capability to tailor the solution to the time scales present in the problem to obtain further reductions in computational cost*

**Flexibility to tailor the method to the time scales of the problem and obtain the required fidelity at a reduced computational cost!**

- **Future work** – methods for obtaining optimal  $p$  and  $r$  and extending the work to models including kinematic nonlinearities and dissipation effects

# THANK YOU FOR YOUR ATTENTION!

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## Main references:

- [1] T. Gail, S. Ober-Blöbaum, and S. Leyendecker, "Variational multirate integration in discrete mechanics and optimal control," *Proceedings of ECCOMAS*, 2017, pp. 1–4.
- [2] O. Junge, J. E. Marsden, and S. Ober-Blöbaum, "Discrete mechanics and optimal control," *IFAC Proceedings Volumes*, Vol. 38, No. 1, 2005, pp. 538–543.
- [3] S. Ober-Blöbaum, O. Junge, and J. E. Marsden, "Discrete mechanics and optimal control: an analysis," *ESAIM: Control, Optimisation and Calculus of Variations*, Vol. 17, No. 2, 2011, pp. 322–352.
- [4] S. Leyendecker and S. Ober-Blöbaum, "A Variational Approach to Multirate Integration for Constrained Systems," *Multibody Dynamics: Computational Methods and Applications* (J.-C. Samin and P. Fiset, eds.), pp. 97–121, Dordrecht: Springer Netherlands, 2013.
- [5] T. Gail, S. Leyendecker, and S. Ober-Blöbaum, "Computing time investigations of variational multirate integrators," *ECCOMAS Multibody Dynamics*, 2013.
- [6] M. Azadi, M. Eghtesad, S. Fazelzadeh, and E. Azadi, "Dynamics and control of a smart flexible satellite moving in an orbit," *Multibody System Dynamics*, Vol. 35, No. 1, 2015, pp. 1–23.
- [7] J. L. Junkins and Y. Kim, *Introduction to dynamics and control of flexible structures*. American Institute of Aeronautics and Astronautics, 1993.